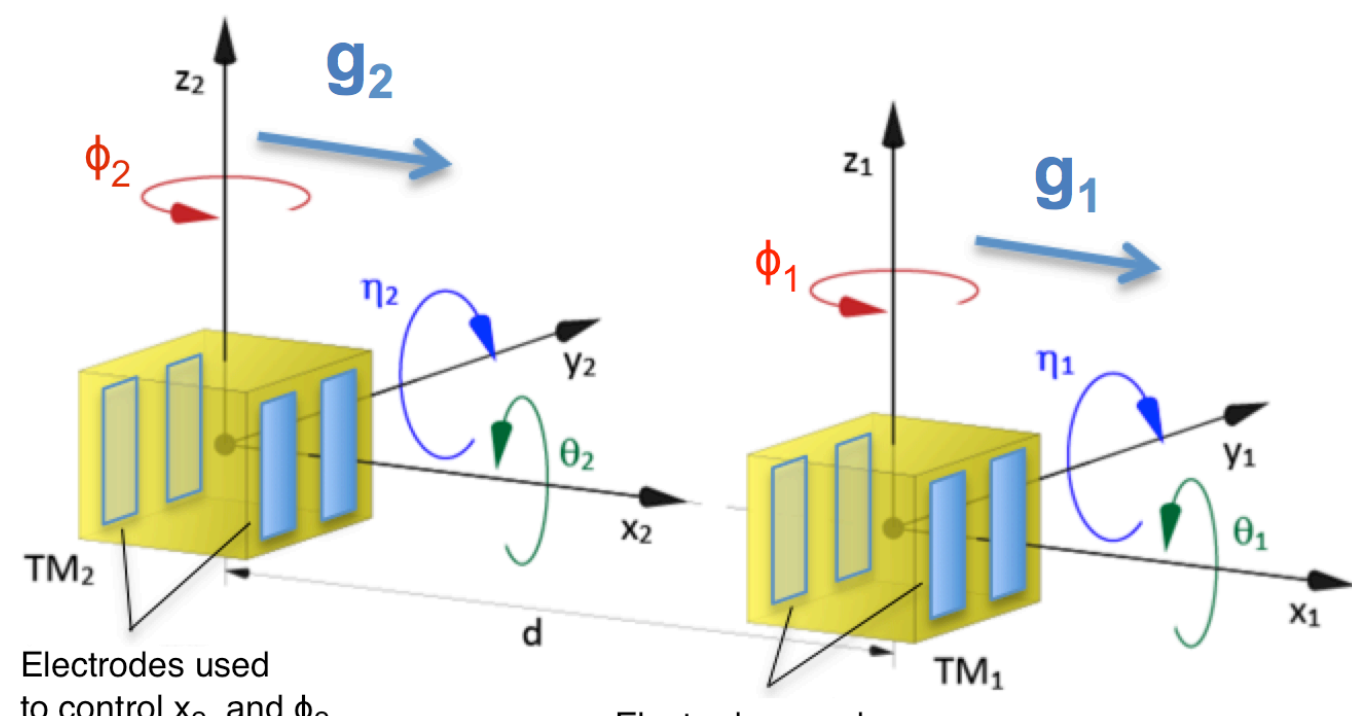


LISA Pathfinder: Free-fall Mode experiment

LPF as an **accelerometer**: measurement of small time-varying accelerations in the presence of a large DC (or very low frequency) acceleration

Compensating such DC/low frequency accelerations with applied forces can limit sensitivity

Compensate for the DC acceleration imbalance → Electrostatic actuation



LPF: 2 reference TM, 1 measurement axis

- Spacecraft cannot follow both TM
- Must force TM2 to follow TM1
- Actuation noise on x axis not representative of LISA and scales with force level
- Force level dictated by spacecraft self-gravity

Fluctuations from electrostatic force component $F \propto V_{ACT}^2$

Actuation force noise contribution if actuating with 1 electrode to compensate Delta g

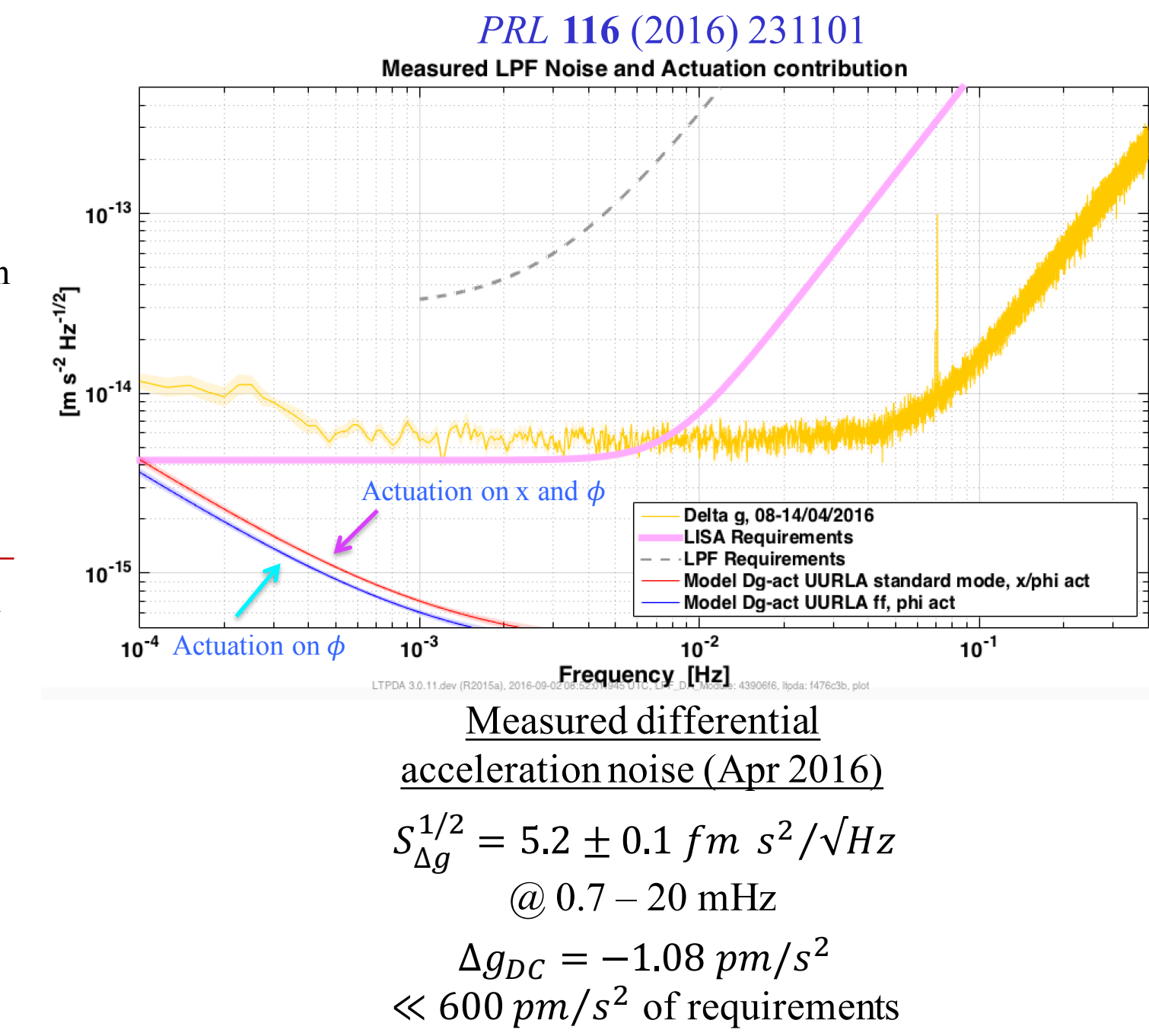
$$S_{\Delta g_{ACT}}^{1/2} \approx 2\Delta g_{DC} S_{\delta V/V}^{1/2}$$

$$S_{\delta V/V}^{1/2} \approx 5 \text{ ppm}/\sqrt{\text{Hz}}$$

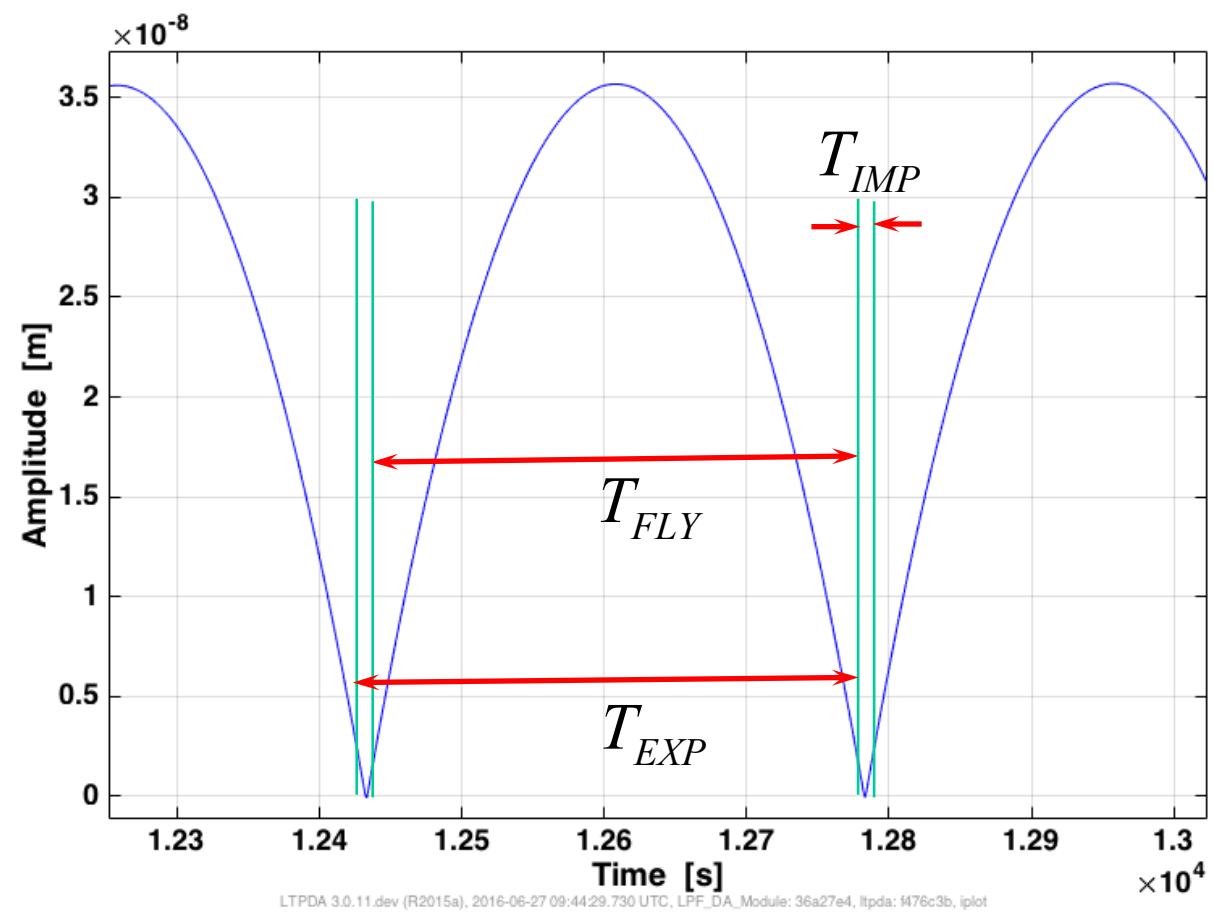
Considering both x / phi forces and torque and suspension authorities:

$$S_{\Delta g_{ACT}}^{1/2} \approx 2\Delta g_{EFF} S_{\delta V/V}^{1/2}$$

$\Delta g_{EFF} \approx 50 \text{ pm/s}^2$
Effective acceleration as a function of torque and forces and suspension authority



Experimental Concept and Implementation



- Compensate force – not continuously – but with series of impulses
- Throw out data during impulses
- Analyze data for the actuation-free «quasi-parabolic» flights between impulses
- Reduced actuation noise and completely independent Delta g calibration

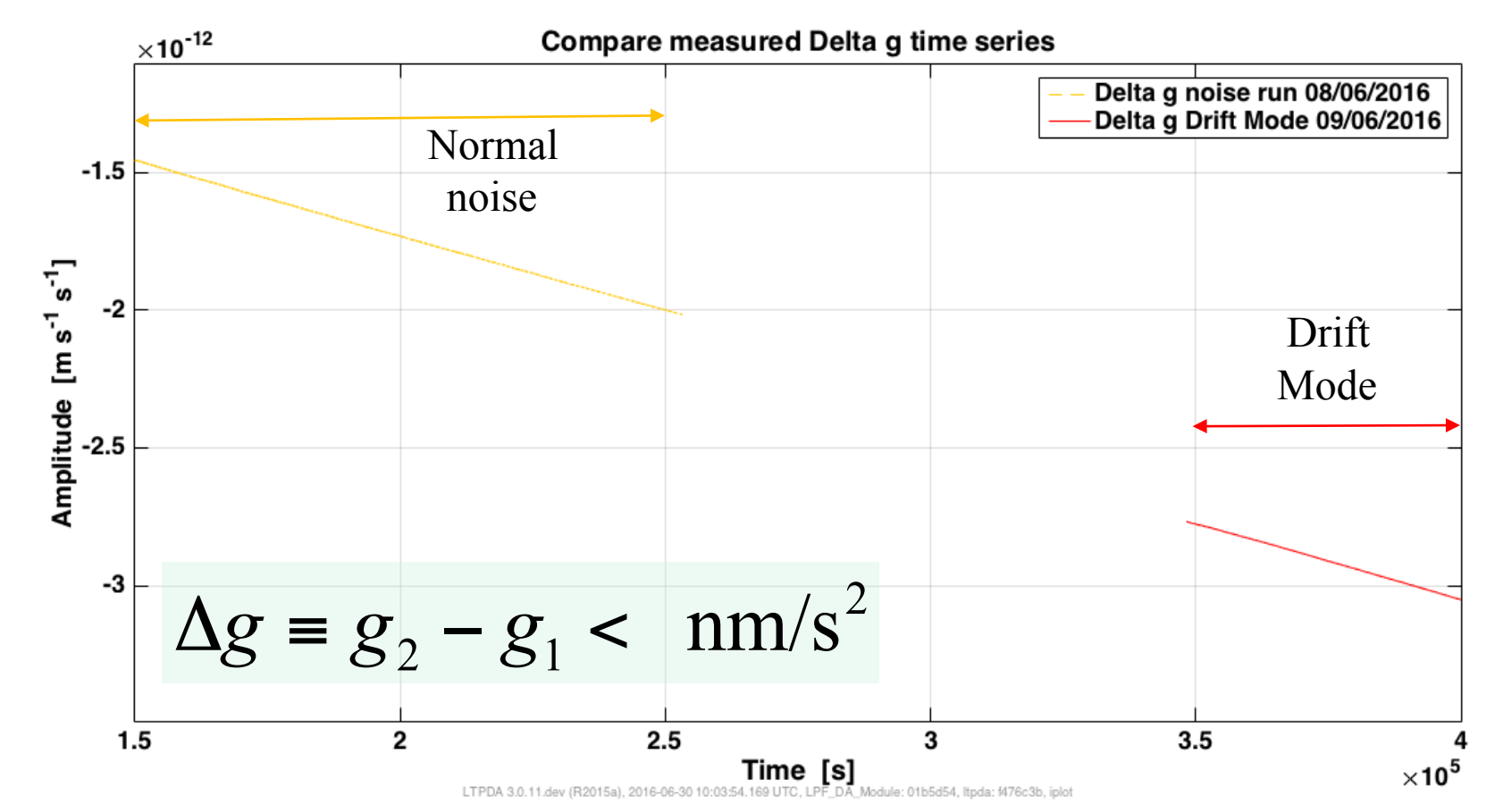
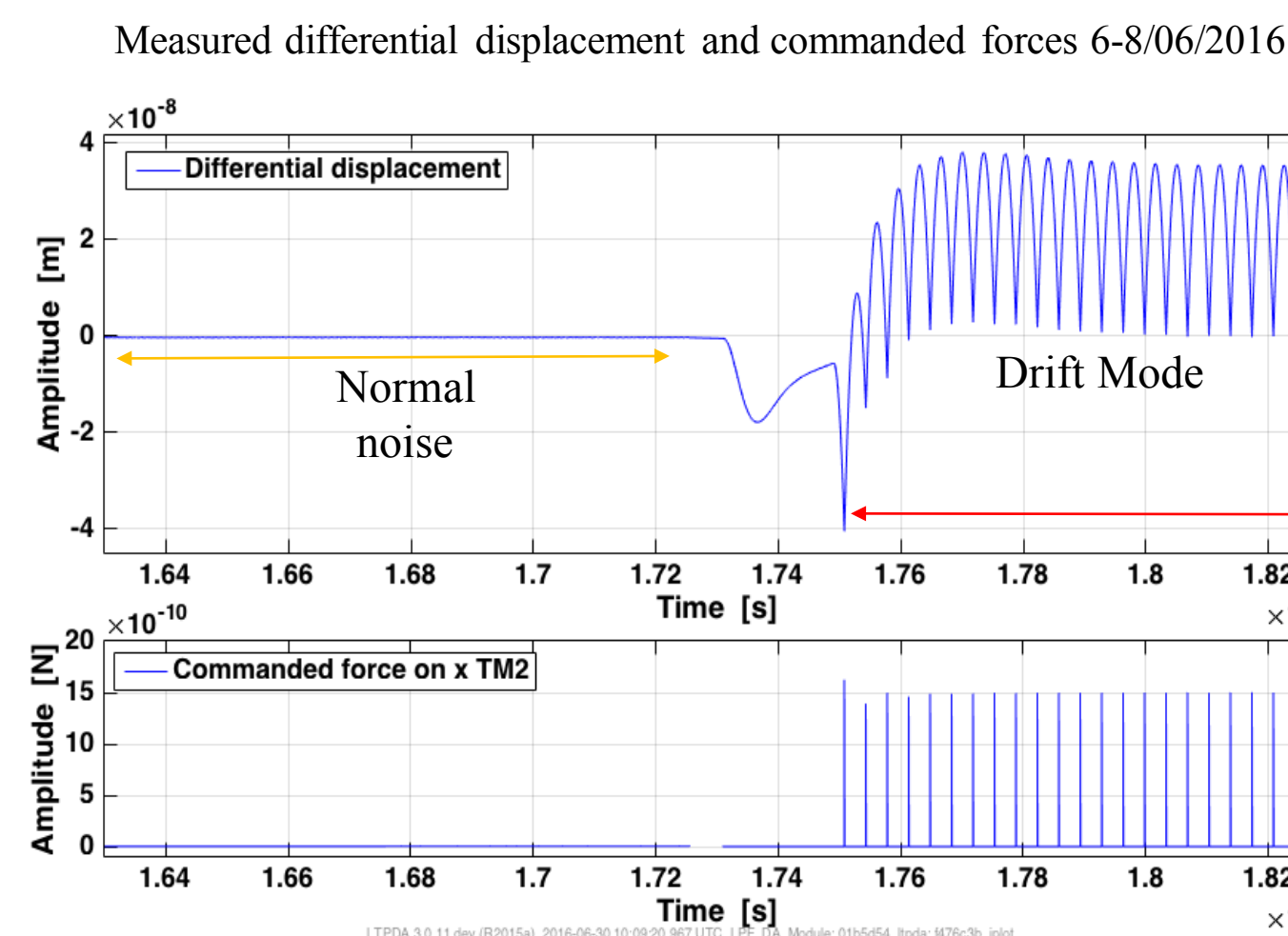
$$\Delta g(t) = \Delta \ddot{x}_{12} + \Delta \omega^2 (x_1 - x_{SC}) + \omega_{2p}^2 \Delta x_{12}$$

Cycle time: $T_{EXP} = 350.2 \text{ s}$
Impulse duration: $T_{IMP} = \chi T_{EXP} = 1 \text{ s}$
Flight time: $T_{FLY} = (1 - \chi) T_{EXP} = 349.2 \text{ s}$
Duty cycle: $\chi = T_{IMP} / T_{EXP} = 0.0028$

Authority x electrodes: 11V High Res., 135V half time Wide Range > drift HR, kick WR

Actuation authority WR: $F_{X1}^{MAX} = 0, F_{X2}^{MAX} \sim 1 \mu\text{N}$
 $\sim 3 \mu\text{N}$ with 0.3% of duty cycle

Applied differential actuation force to compensate self gravity difference $f = m\Delta g_{DC} \approx 2 \text{ nN}$ with $|\Delta g_{DC}| \approx 3 \text{ pm/s}^2$



Measured time series (9-10/06/2016)

- 2 days of drift mode in two actuation configurations
- # flights 453
- Avg Drift len. = 349.2 s, Avg kick len. = 1 s
- Avg Kick Amplitude = 1.94 nN
- Flights amplitude ~ 35 nm
- Sensitive only to the interferometer noise and not to the actuator calibration during free phases

Operational questions:

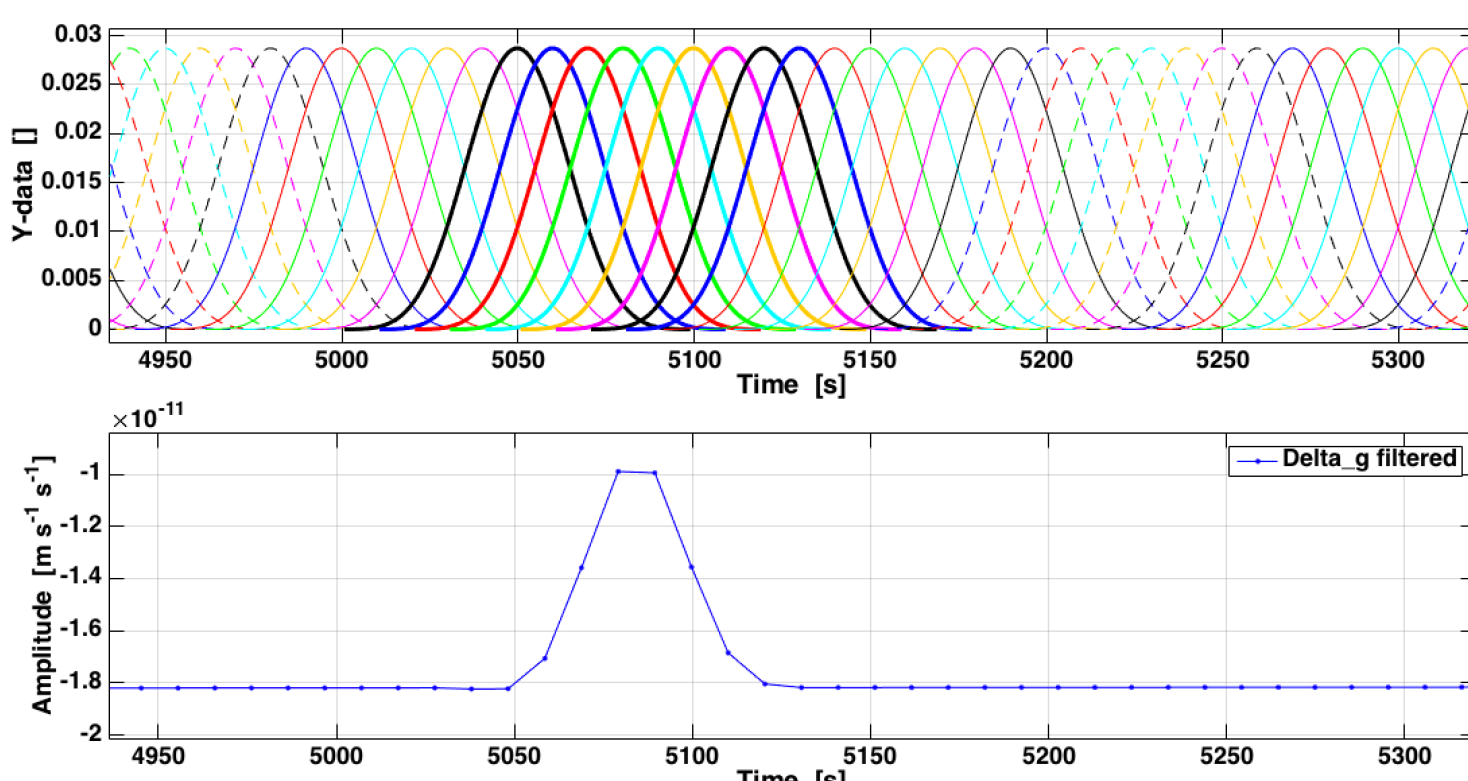
- Can we recover true «actuation-free» acceleration noise at $\text{fm/s}^2/\text{Hz}^{1/2}$ level:
 - with gaps in data?
 - with much larger displacement dynamic range?

Data analysis techniques

Blackmann-Harris technique

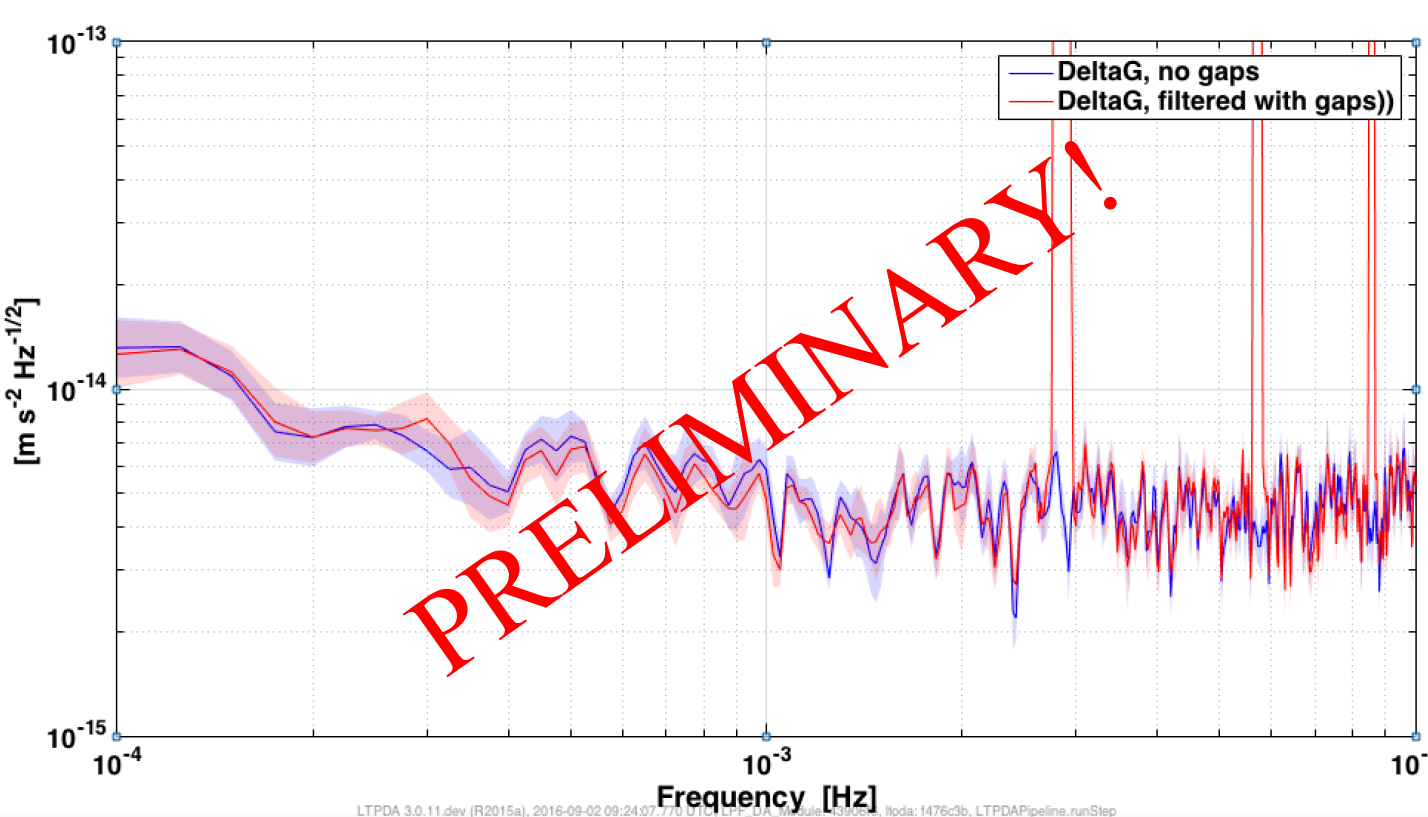
Approach description:

- Low pass filter with a Blackmann-Harris window normalized to give unit transfer function at $f=0$
- Data decimated of a factor 103 (sample time: 10.3 s), put to zero data in the gap
 - Total number of samples per flight: $n_{tot} = 34$
 - Number of samples in the gap: $n_{keep} = 9$
- Length of window: $T_{win} = T_{fly} - 2 \cdot \text{margin} - (n_{keep} - 1)T_{samp} \sim 98 \text{ s}$



- Technique applied on a standard science noise run
- PSD corrected for gap ratio (n_{tot}/n_{keep}) and BH filter transfer function
 - Bias is model dependent
 - Still under consolidation

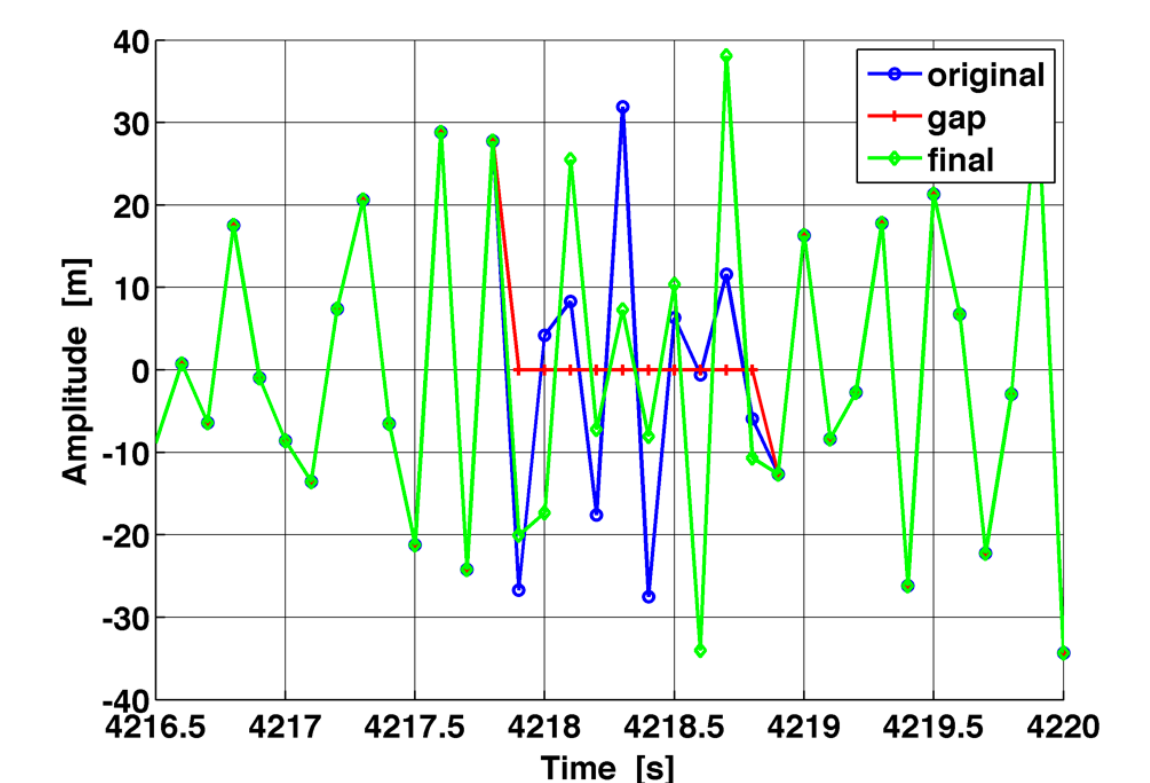
Noise test



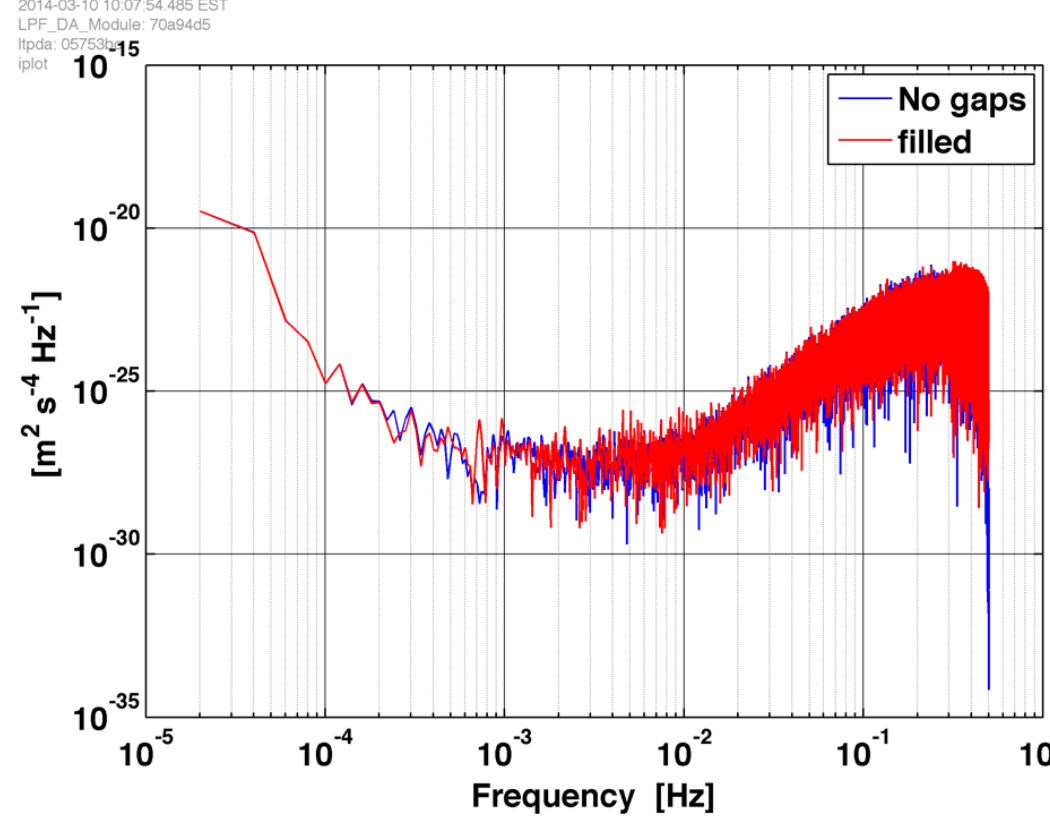
Constrained gaussian gap - patching

Approach description:

- Fill gaps with random data generated from an assumption of an underlying noise spectrum
- **Algorithm**
 - Create two-point function from model spectrum
 - draw zero-mean random samples matching that distribution
 - Use data adjacent to gaps and two-point function to adjust mean of random samples
 - Update spectral model and iterate if necessary



Noise test



Bias of the technique

Comparison of best-fit parameters for spectral model $S_{mod}(f) = P_{-6} \cdot f^{-6} + P_{-2} \cdot f^{-2} + P_0 \cdot f^0 + P_4 \cdot f^4$ for original and CG patched spectral densities.

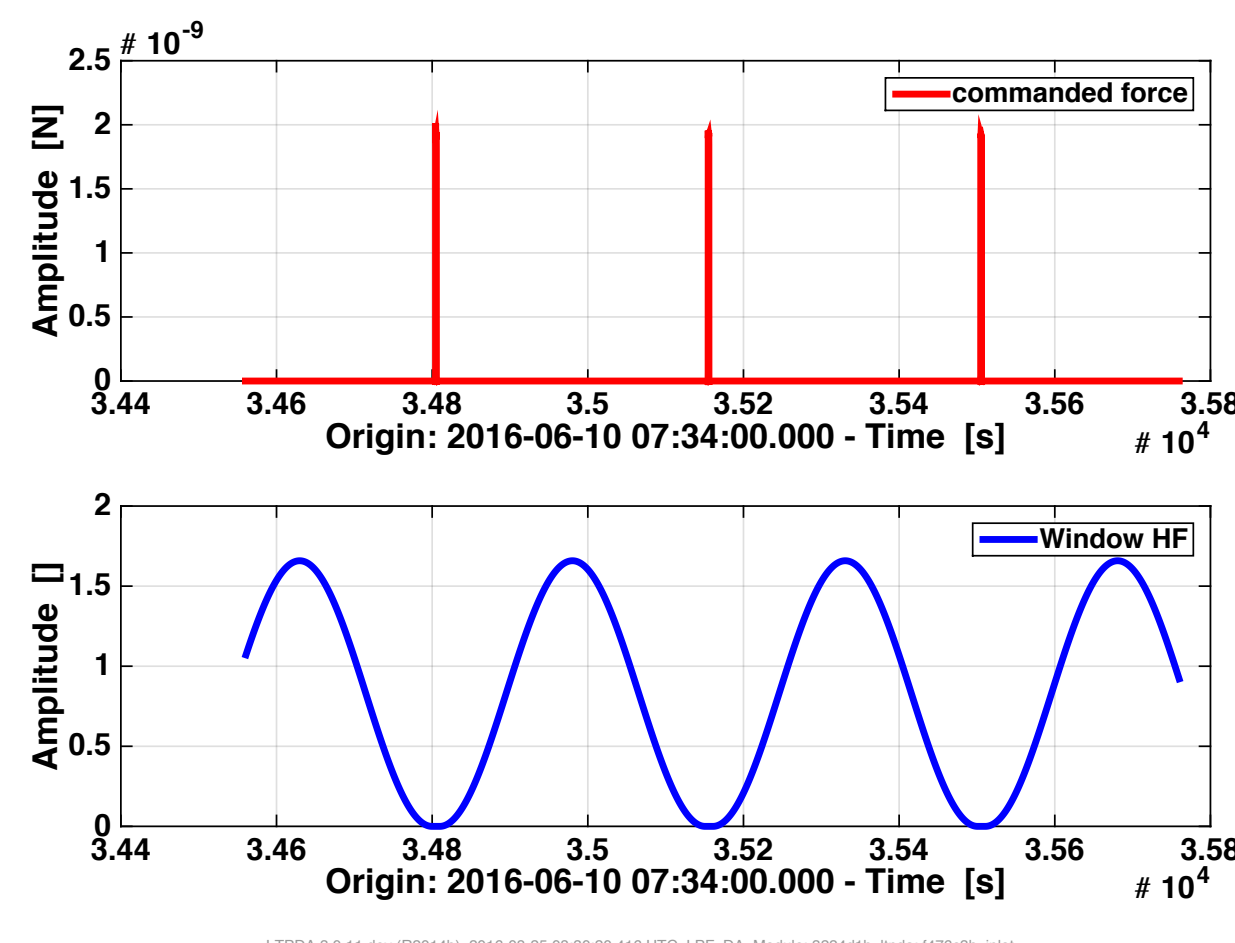
	original	μ	σ	CG patching	μ	σ	$\frac{\mu - \text{theory}}{\sqrt{\sigma^2 + \sigma_{theory}^2}}$
$P_{-6} \times 10^{20}$	0.66	0.66	0.66	0.71	0.03		
$P_{-2} \times 10^{20}$	1.14	0.18	1.08	0.20	-0.2		
$P_0 \times 10^{20}$	2.25	0.16	2.31	0.19	0.3		
$P_4 \times 10^{20}$	6.23	0.03	6.22	0.03	0.09		

Journal of Physics: Conference Series 610 (2015) 012006

Windowing method

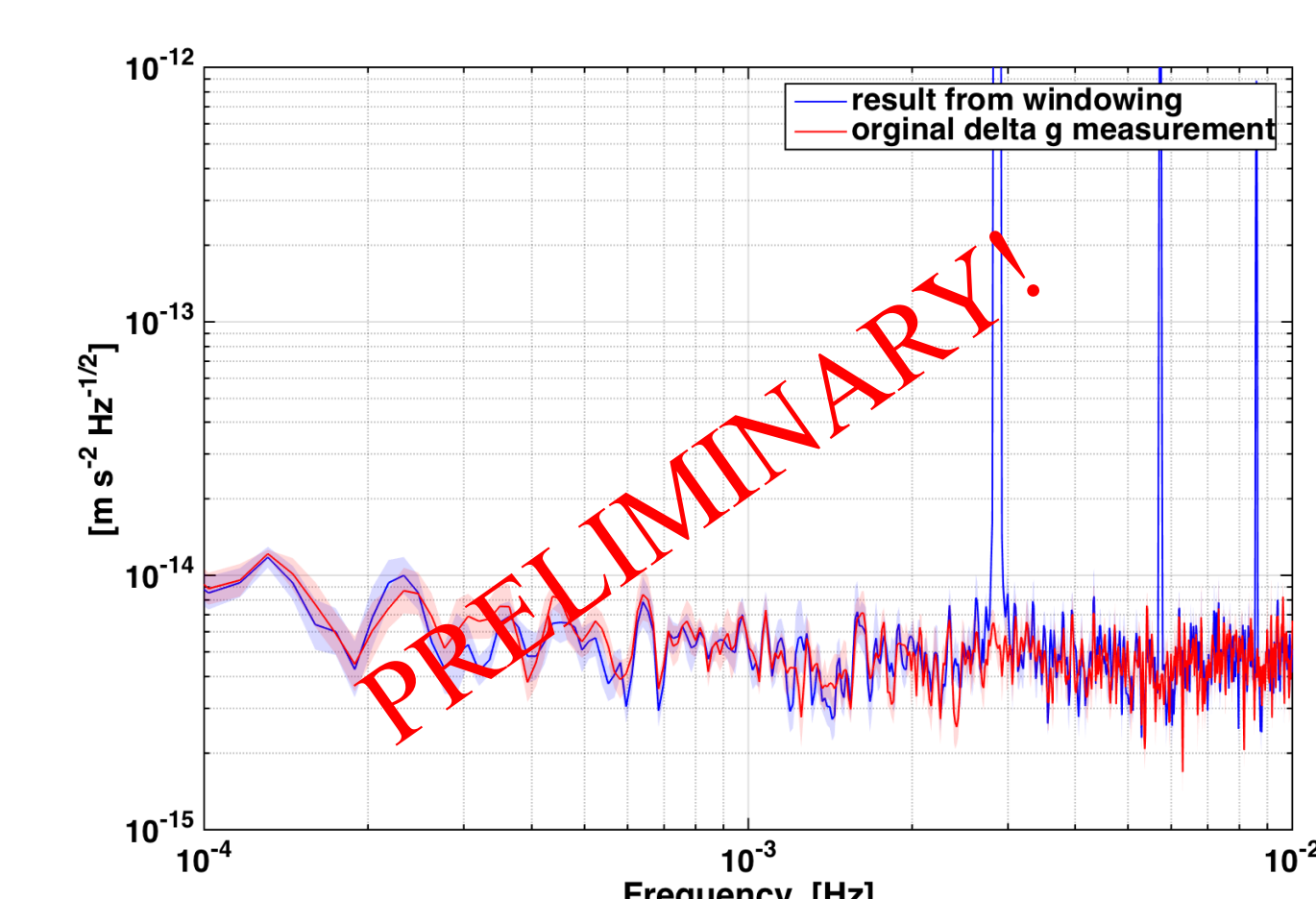
Approach description:

- do a simplex fit using an adapted logarithmic likelihood function on the non-windowed data and verify results with MCMC
- also fit DC acceleration to remove the effect of the window $\ddot{o}_{12} = -\omega_1 \ddot{o}_{12} + g_{12} + \dot{g}_0 t$
- Build a customized window that smoothly goes to zero at each gap location (suppress gap transfer function)
 - note: window is normalised such that no spectral leakage from white noise occurs
- Apply normal PSD estimation techniques



- Technique applied on a standard science noise run
- Bayesian model selection is pending
- Known bias in method of up to 20% (underestimate) at 0.1 mHz

Noise test



Comparisons and results

Calibration of the experiment:

- Low pass data with a Blackmann-Harris (~ 60 mHz)
- Fit to each flight $\ddot{o}_{12} = -\omega_{12} \ddot{o}_{12} - \omega_{12} \ddot{o}_{12} + g_{12} + \dot{g}_0 t$
- Evaluate the Delta g to analyze with different techniques

Parameter	mean value	error
Diff stiffness [s^{-2}]	-1.2707e-07	4.0704e-08
Stiffness 2 [s^{-2}]	-4.5856e-07	1.8691e-09
Diff DC Acc [m s^{-2}]	-3.0295e-12	8.9170e-15
Acceleration variation [$\text{m s}^{-2} \text{D}^{\wedge} 1$]	-4.9304e-13	1.5688e-14

ω_2 prediction from electrostatic = [-4.308 ± 0.05] e-07 s-2

