

Program in detection of gravitational wave in space in Chinese Academy of Sciences

Gang JIN

On behalf of Space Gravitational Wave
Detection Working Group in Chinese
Academy of Sciences

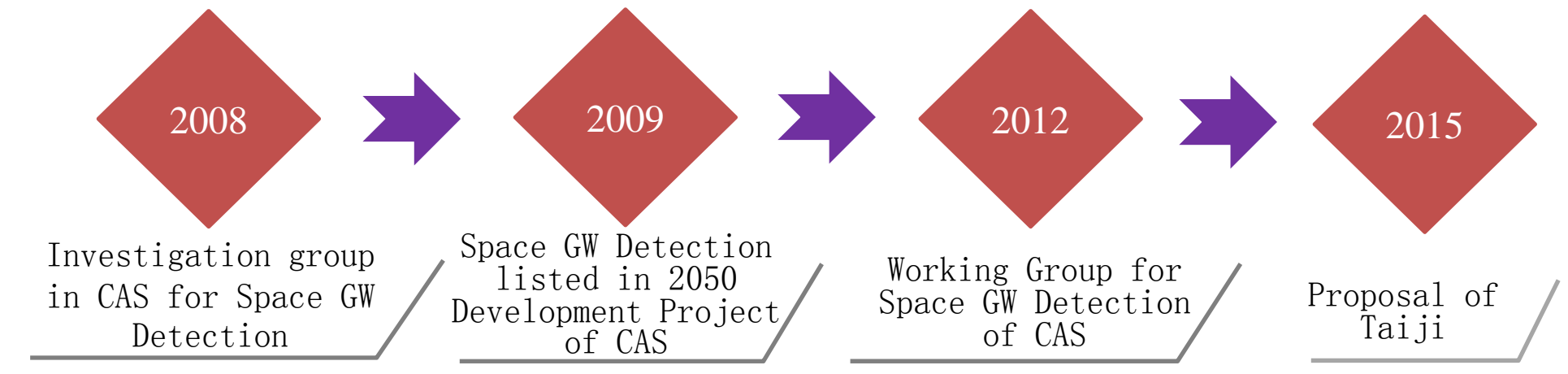
11th LISA Symposium
Zurich, Sept.5-9, 2016



Outline

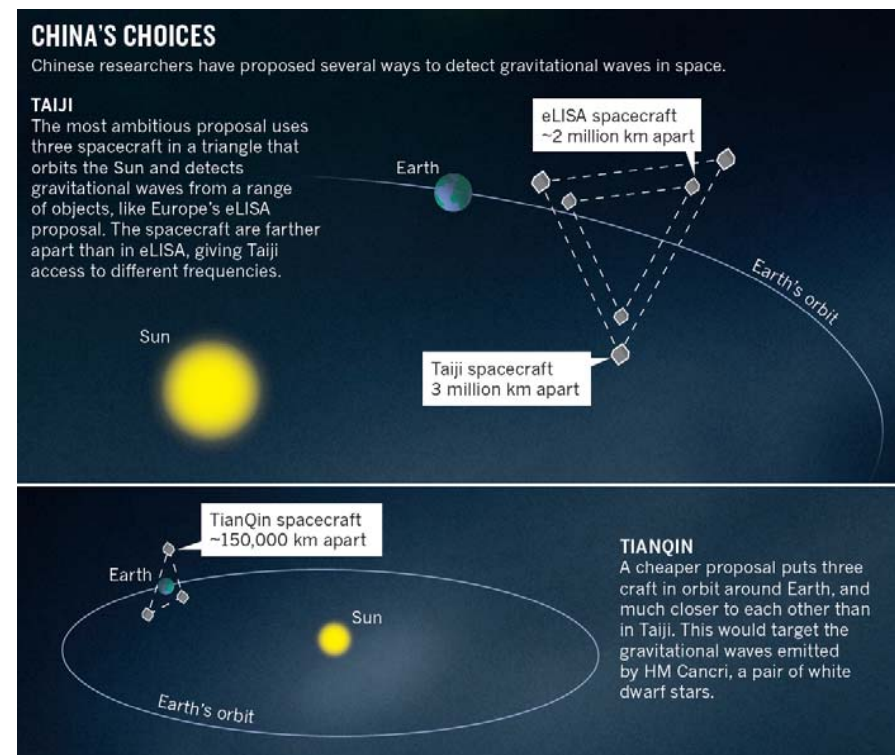
- Roadmap of gravitational wave detection in Space in Chinese Academy of Sciences
- Primary science drivers
- Preliminary mission design
- Technological developments
- Some international activities

Progress of Space gravitational wave detection in China



Taiji proposed and supported in 2016 by Chinese Academy of Sciences, uses three Spacecraft in a triangle that orbits the Sun.

Tianqin proposed by Zhongshan Univ. and supported by local government, orbits the Earth.



Taiji gravitational physics consortium in CAS

Coordinators:

Wenrui Hu, Gang Jin (National Microgravity Laboratory, Institute of Mechanics)

Yueliang Wu (University of Chinese Academy of Sciences)

●Member Institutes participating in the group:

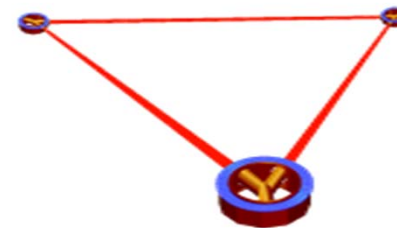
- Academy of Mathematics and Systems Science,
- Changchun Institute of Fine Optics and Engineering,
- Institute of Applied Physics, Lanzhou, CAST,
- Institute of high Energy Physics,
- Institute of Mechanics,
- Institute of Physics,
- Institute of Theoretical Physics,
- Nanjing Institute of Astronomy and Optics,
- National Astronomical Observatory,
- Shanghai Small Satellite Company,
- University of Chinese Academy of Sciences,
- University of Science and Technology of China, Hefei,
- Wuhan Institute of Physics and Mathematics,
- Wuhan institute of Geodesy and Geophysics.



中国科学院
CHINESE ACADEMY OF SCIENCES

Roadmap of GWD in Space in CAS

- 2016-2020:
Technique Prototype developments and ground testing.
- 2021-2025:
Technological developments and a pathfinder mission
Two options (or both) to be considered (gravity satellite and a long interferometry arm in deep space)
- 2026-2035:
Chinese mission for gravitational wave detection in Space and launching



Chinese Pathfinder in planning



- Deep space (likely to be L_1 or L_2)
- 10^5 km apart
- Apart from inertial sensor, also test single arm optical interferometry (telescope, pointing, metrology,etc)

Dual tracks of development in GW mission



Develop a Chinese Mission
with international
collaboration

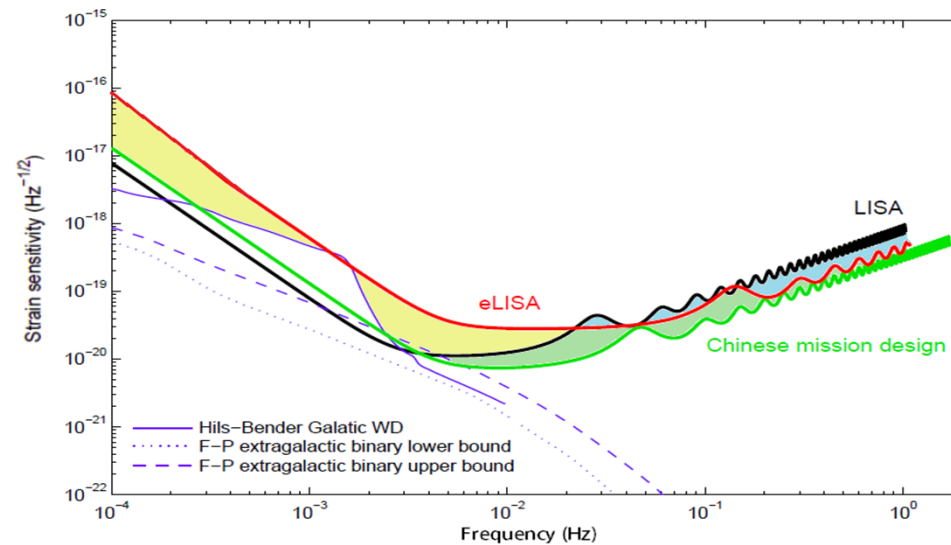


Contribute to eLISA
or LISA likely to be > 20%

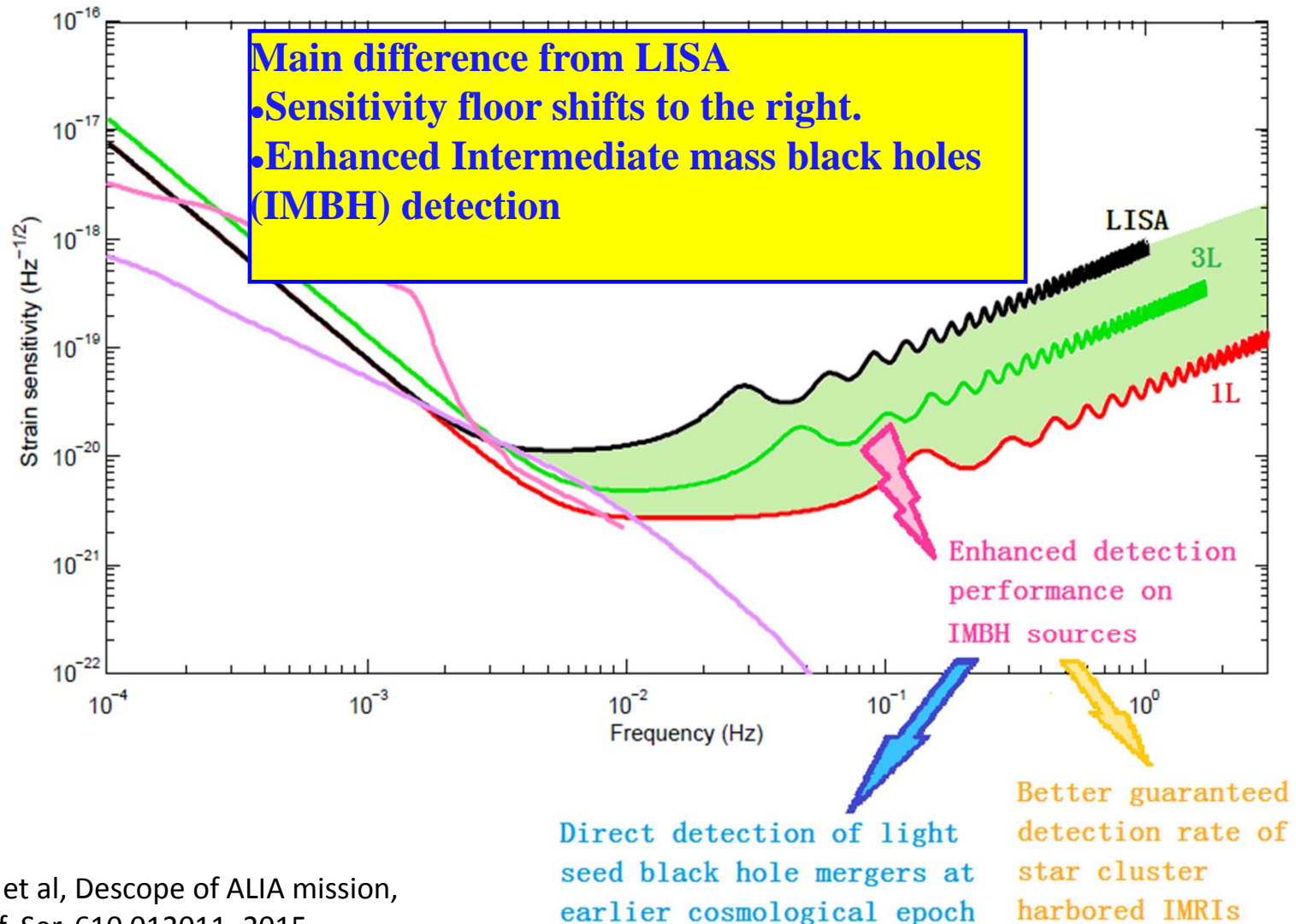
- Micro-Thruster
- Telescope,
- Laser,
- Launcher,
- others ...

Chinese GWD in Space Mission design

Armlength (m)	Telescope diameter (m)	Laser power (W)	1-way position noise ($\frac{\text{pm}}{\sqrt{\text{Hz}}}$)	Acceleration noise ($\frac{\text{m s}^{-2}}{\sqrt{\text{Hz}}}$)
3×10^9 (Chinese mission option)	0.46	2	8	3×10^{-15} (> 0.1mHz)
1×10^9 (eLISA)	0.2	2	11	5×10^{-15} (> 0.1mHz)
5×10^9 (LISA)	0.4	2	18	3×10^{-15} (> 0.1mHz)



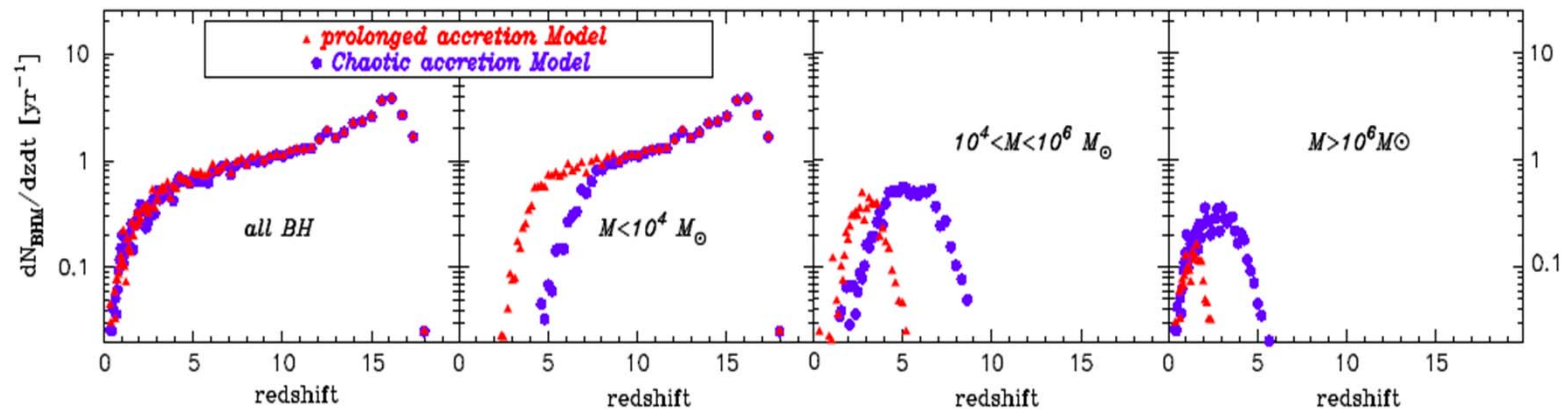
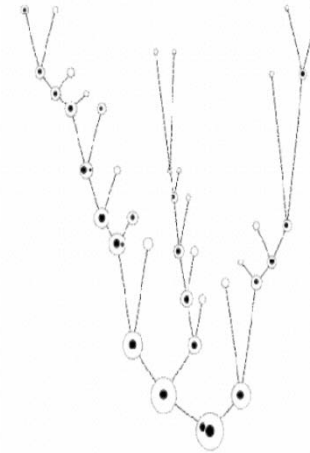
Enhanced detection performance on IMBH



Event rate estimates of MBH mergers at earlier cosmological epoch

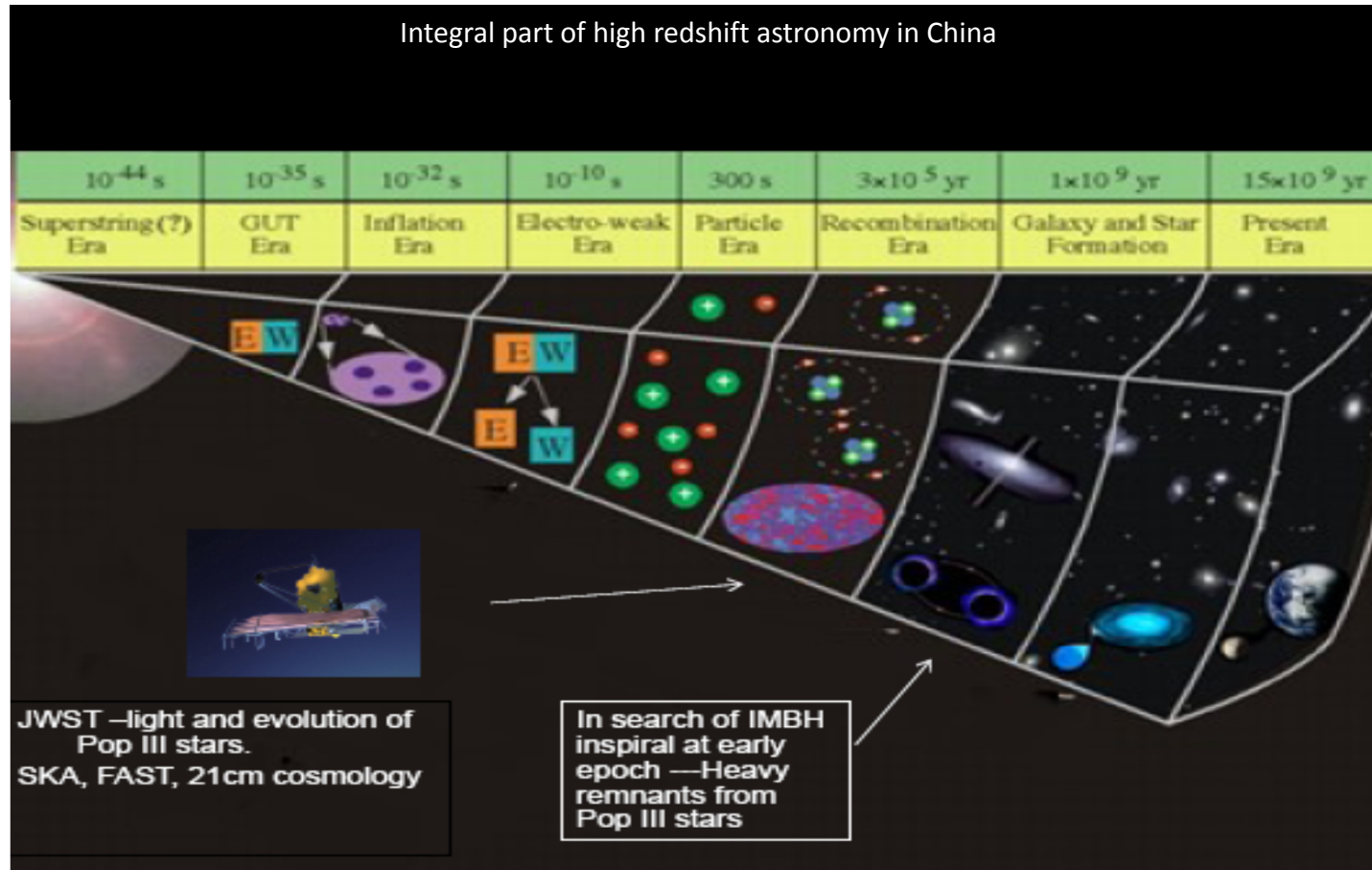
Cosmological MBH merger simulation based on

- Monte Carlo realization of EPS formalism
- Equal mass light seed black hole (PopIII remnants) seeding
- Semi-analytical dynamics ———
 - prolonged and chaotic accretion models
 - coalescence spin and recoil determination:
 - numerical relativity fitting formula



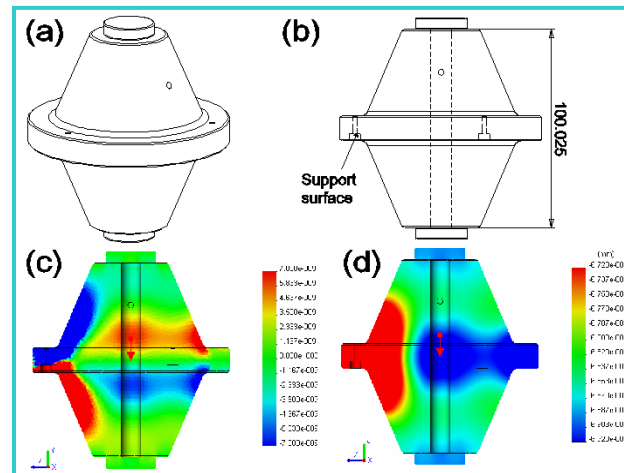
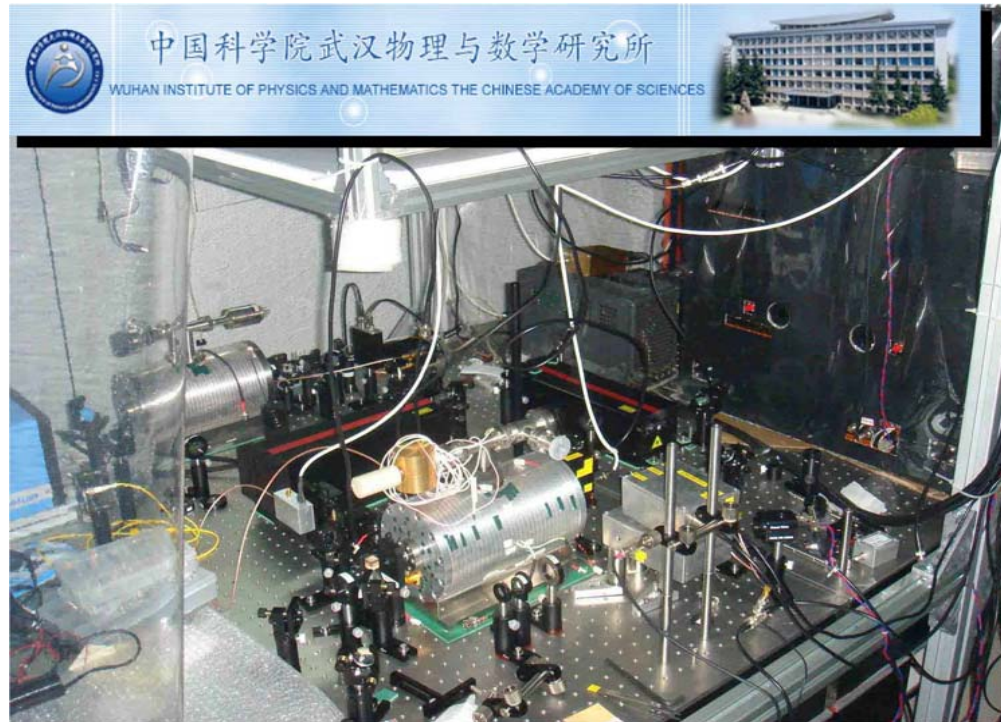
Coalescence rate predicted by the realized simulations.

Astronomy application



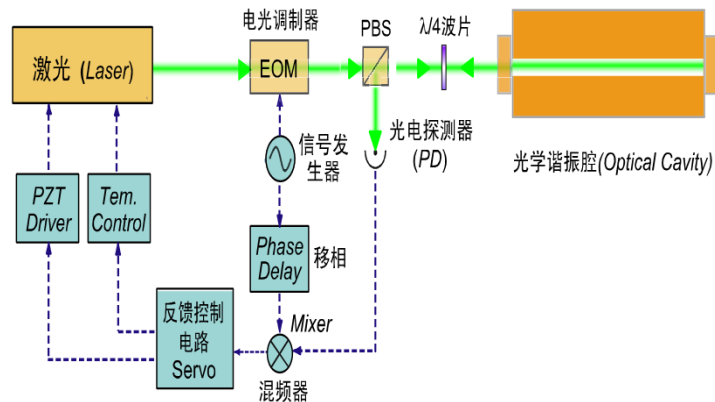
Taiji mission is going to be an integral part of high redshift astronomy in China, complementary to the Infrared and radio astronomy.

Experiments on frequency locking of Nd:YAG lasers



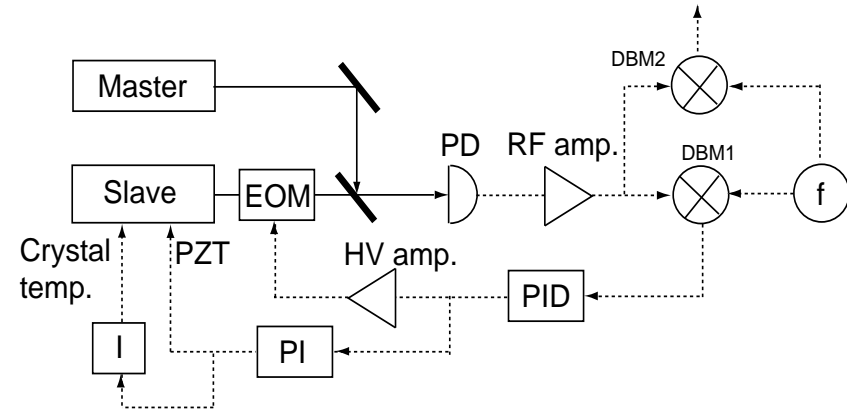
Ongoing experiments in the Wuhan Institute of Physics and Maths, CAS

Frequency locking of Nd:YAG lasers



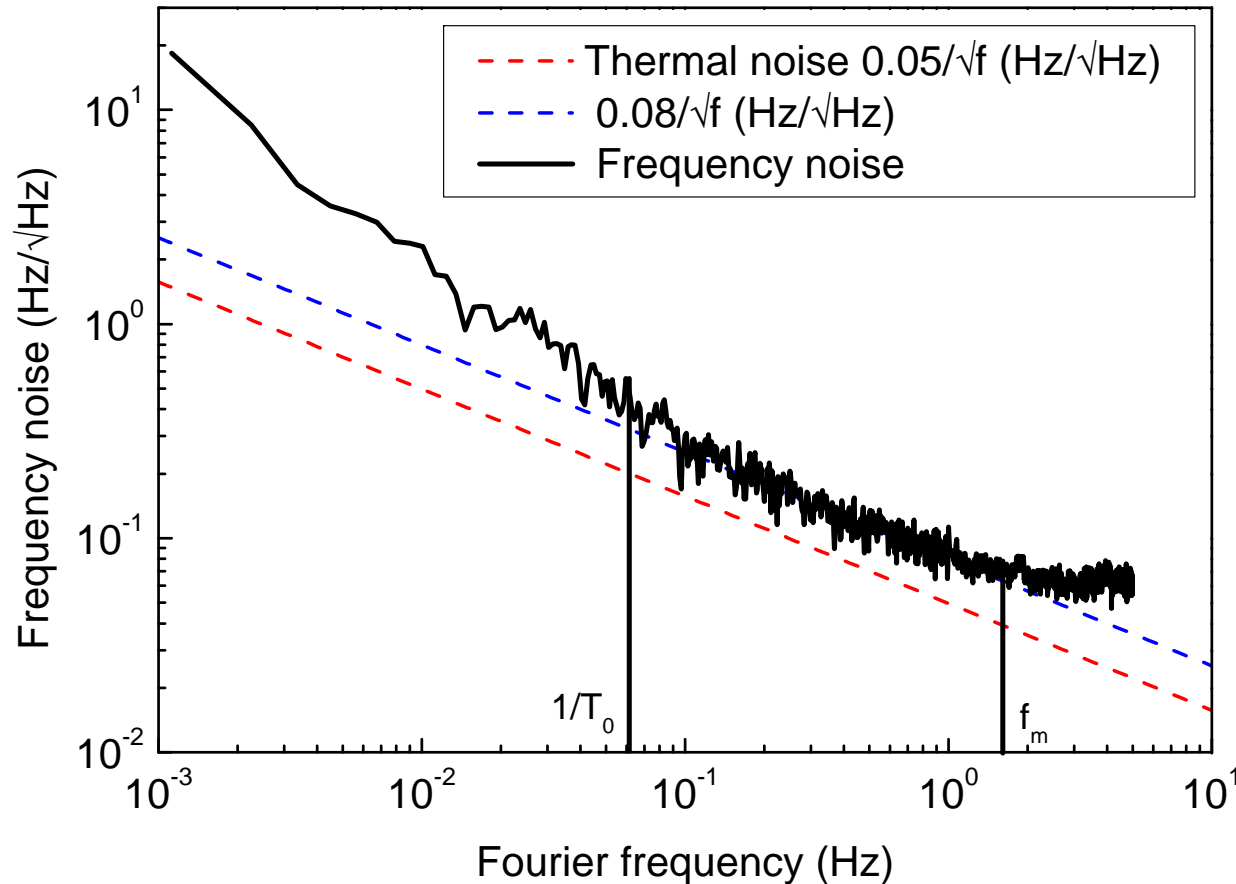
- Projected system performance:
- Frequency noise: $30 \text{ Hz}/\sqrt{\text{Hz}}$ (10mHz-100mHz),
- Frequency drift: $<10 \text{ kHz}/\text{hour}$

Phase lock of two Nd:YAG lasers



- Analog PLL(Phase Lock Loop) experiments
 - Projected phase noise: $1 \times 10^{-4} \text{ rad}/\sqrt{\text{Hz}}$ (10 - 100 mHz)
- Digital PLL – design and experiments
 - a.digital phase detector
 - b.analysis of phase noise
- Our goal: Increase the gain at low frequencies (1 mHz-1 Hz) while extend the unit-gain frequency up to several MHz.

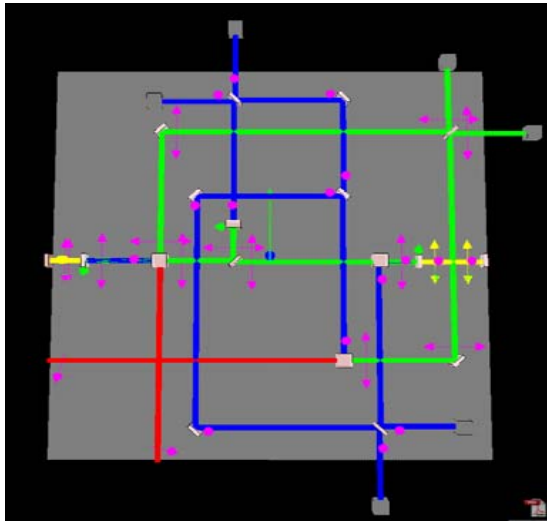
Frequency noise of two beating Nd:YAG lasers independently referenced to 20 cm cavities



Laser linewidth deduced from frequency noise:
0.11 Hz (FWHM) (assuming a measurement time of $T_0=16$ s)

Notice that the above result is achieved in a ground-based system. One should bear in mind that the effort to reach the same stability for a spaceborne laser are challenged by many new technical difficulties.

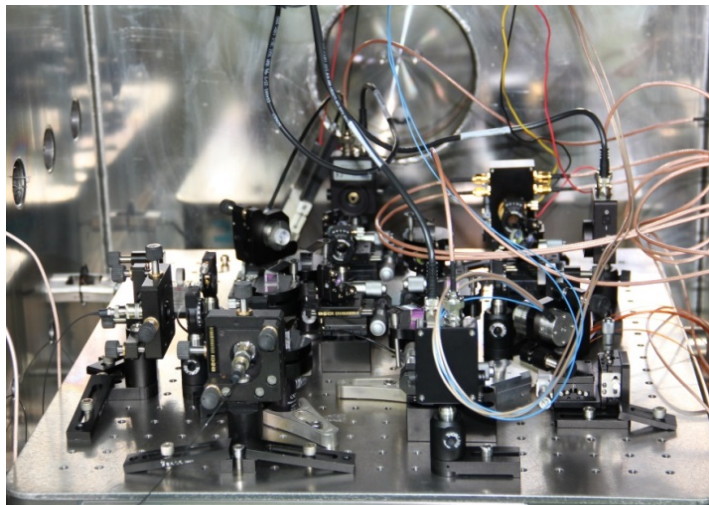
Laser interferometer prototype in I.Mech



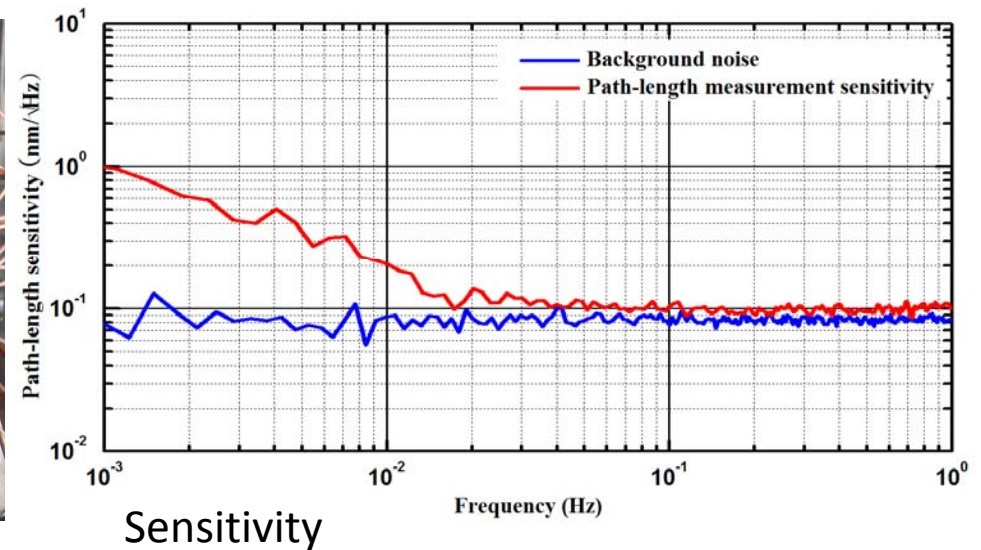
Simulation of optical path



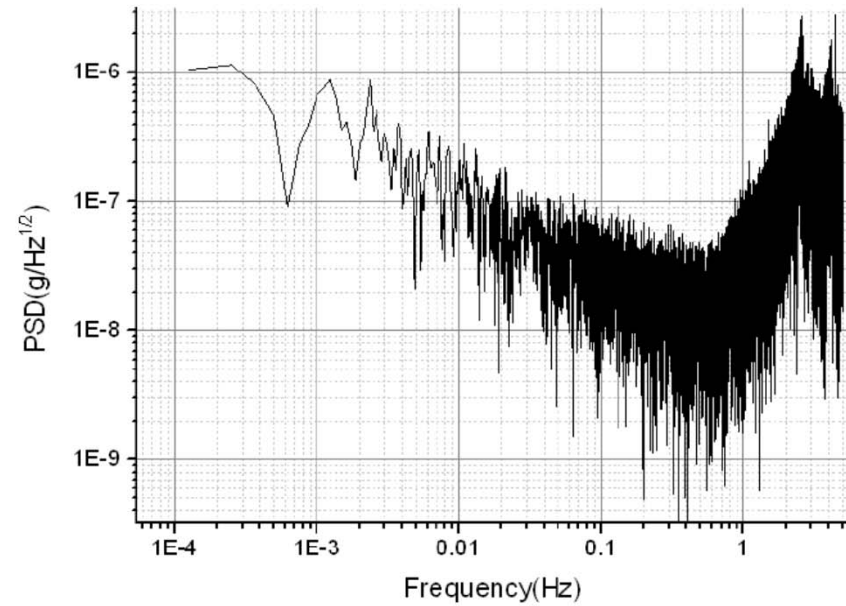
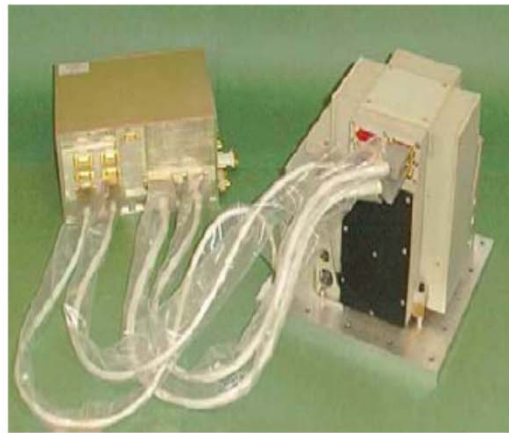
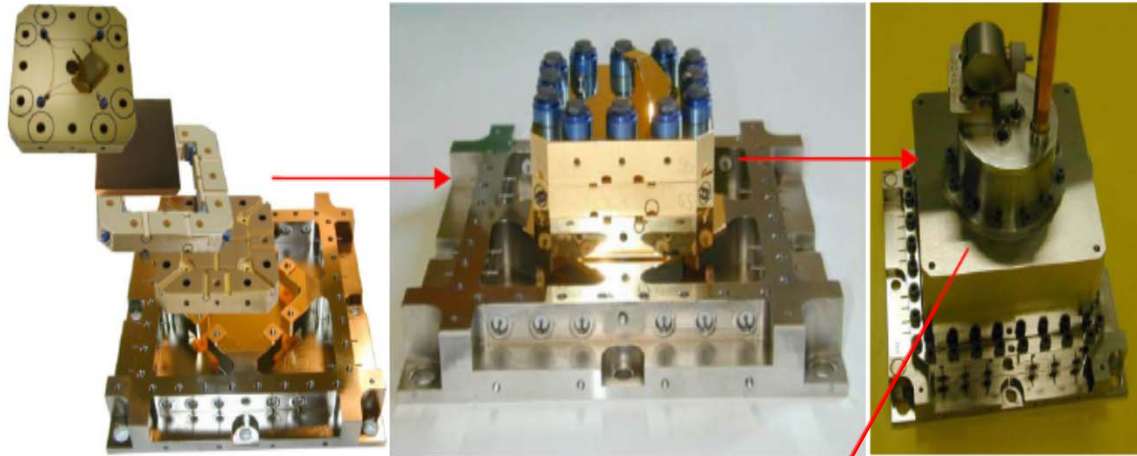
Experimental setup



Interferometer



Accelerometer in Lanzhou Institute of Physics, CAST

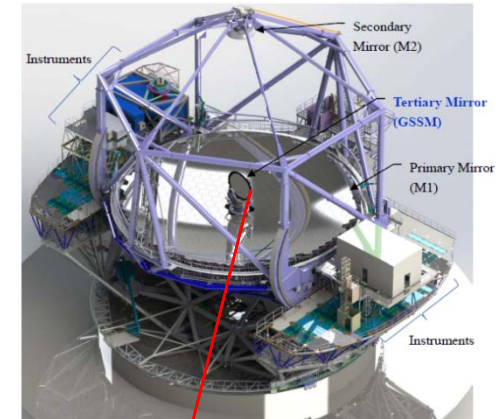


Telescope in Changchun Institute of Optics, Fine Mechanics and Engineering, CAS

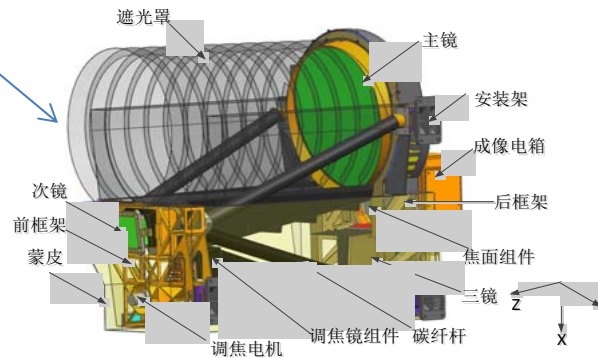
Expertise in off axis triple mirror assembly



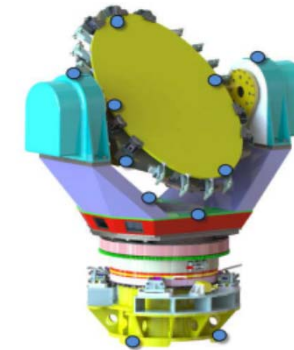
SiC mirror of 4 m aperture



Thirty Meter Telescope (TMT), USA

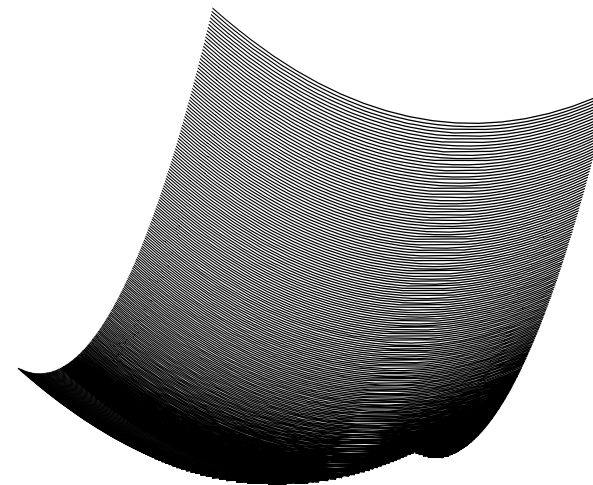
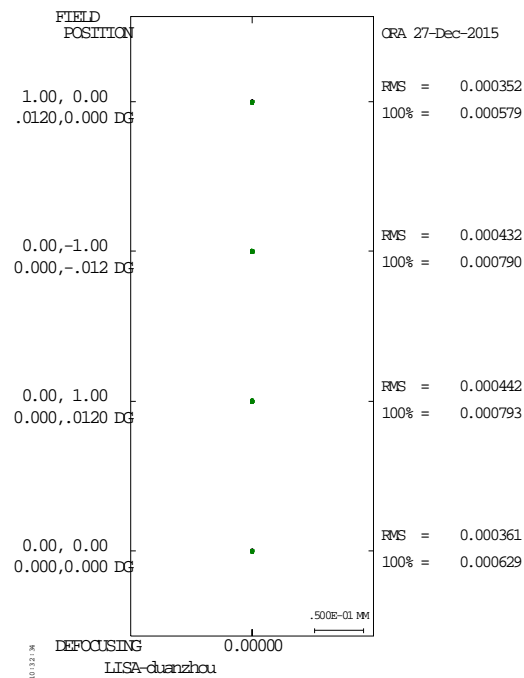
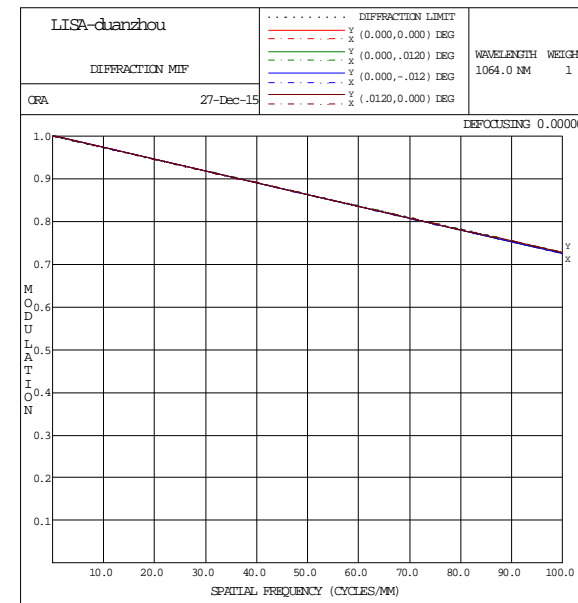
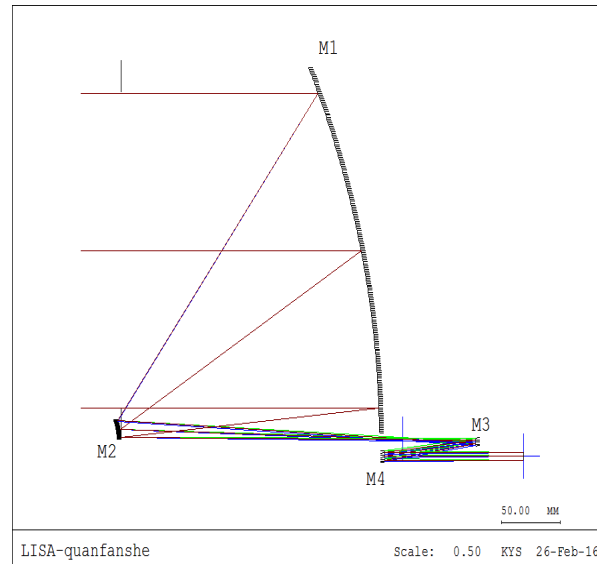


High resolution telescope of Mars 2020

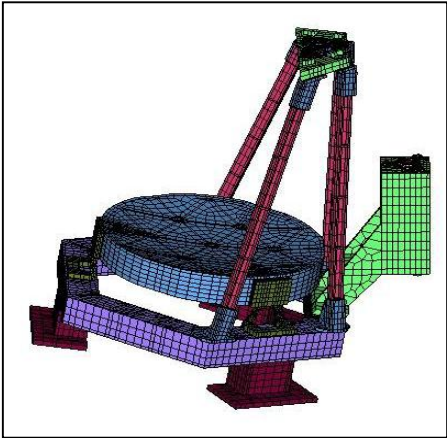
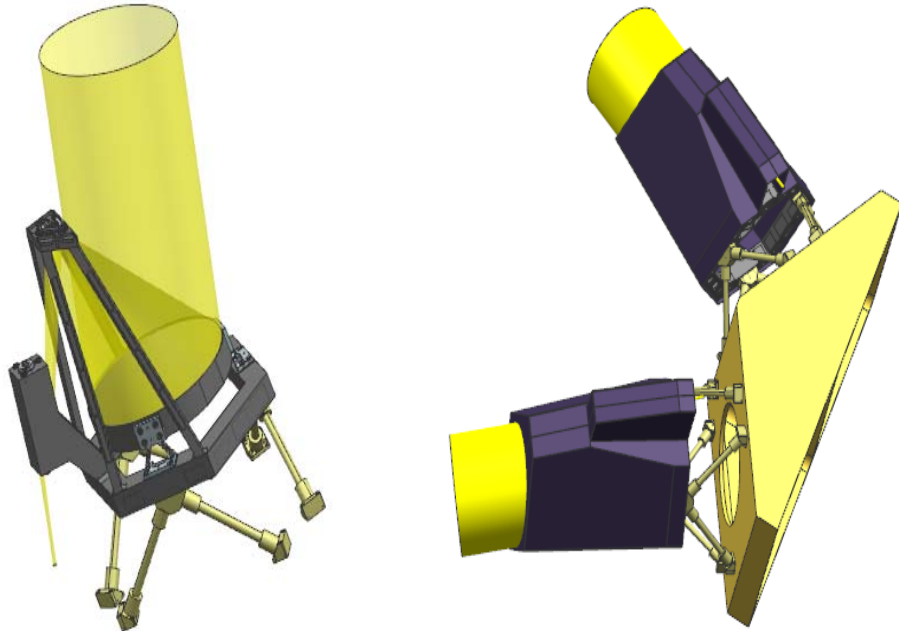


Tertiary Mirror subsystem, designed by CIOMP and aperture is $3\text{m} \times 2.5\text{m}$.

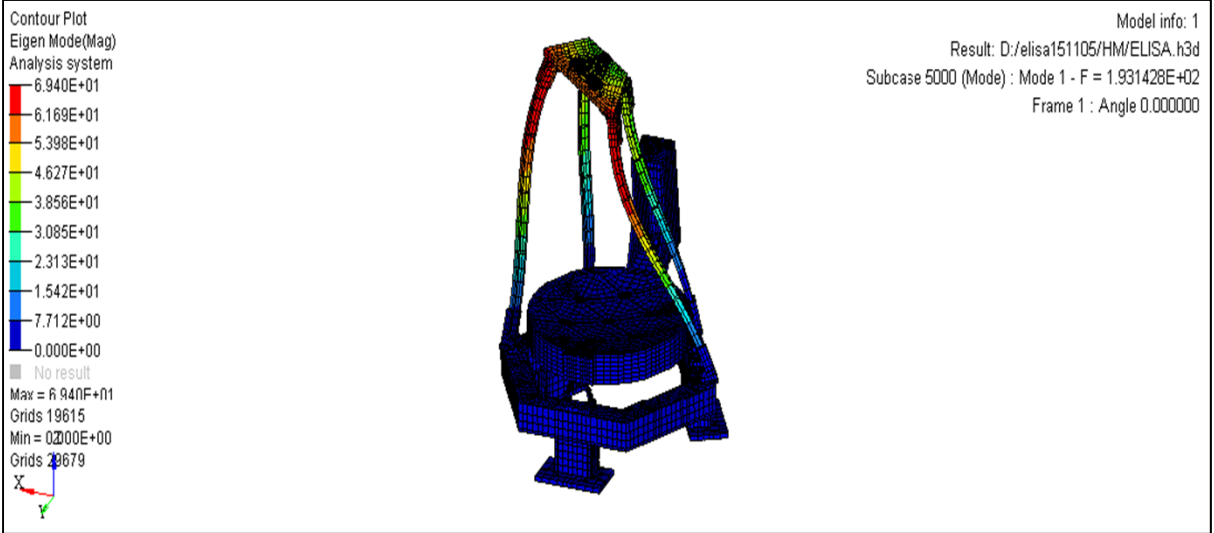
Optical system design and analysis of LISA telescope



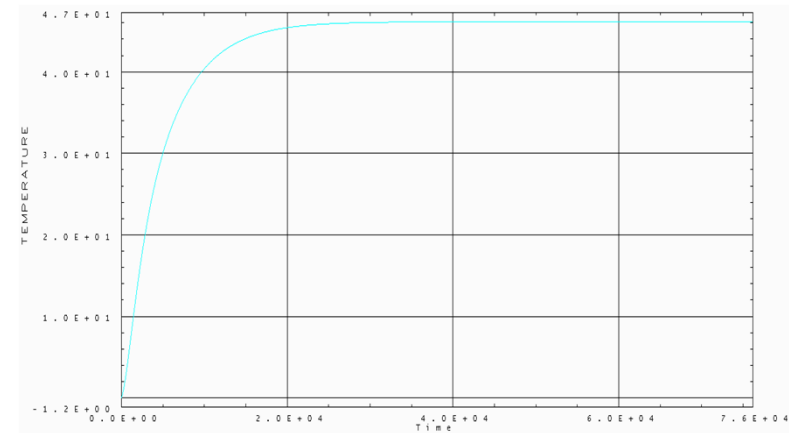
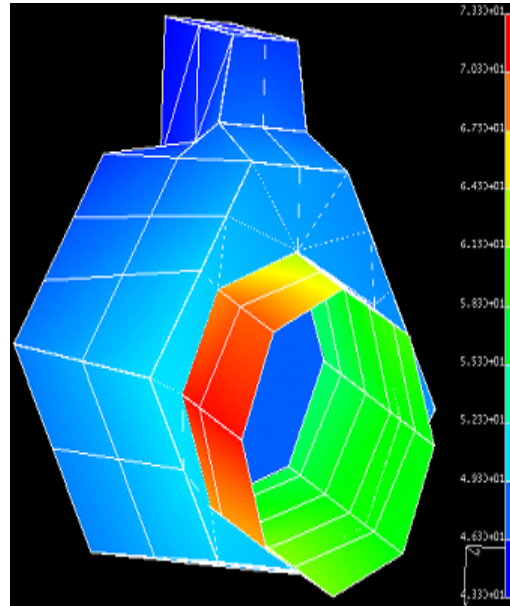
Structure design and analysis of eLISA telescope



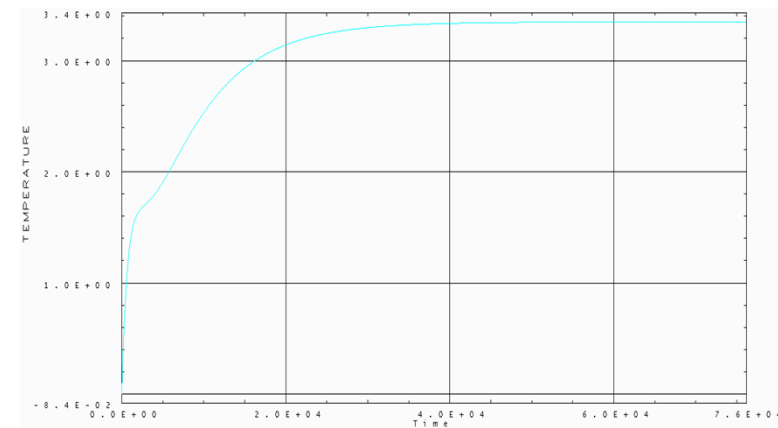
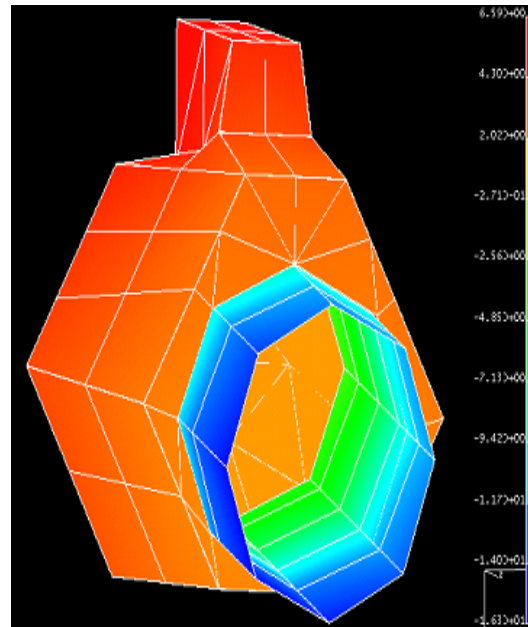
First-order modal: 193Hz
Second-order modal: 301Hz
Third-order modal: 376Hz



Preliminary thermal analysis of telescope

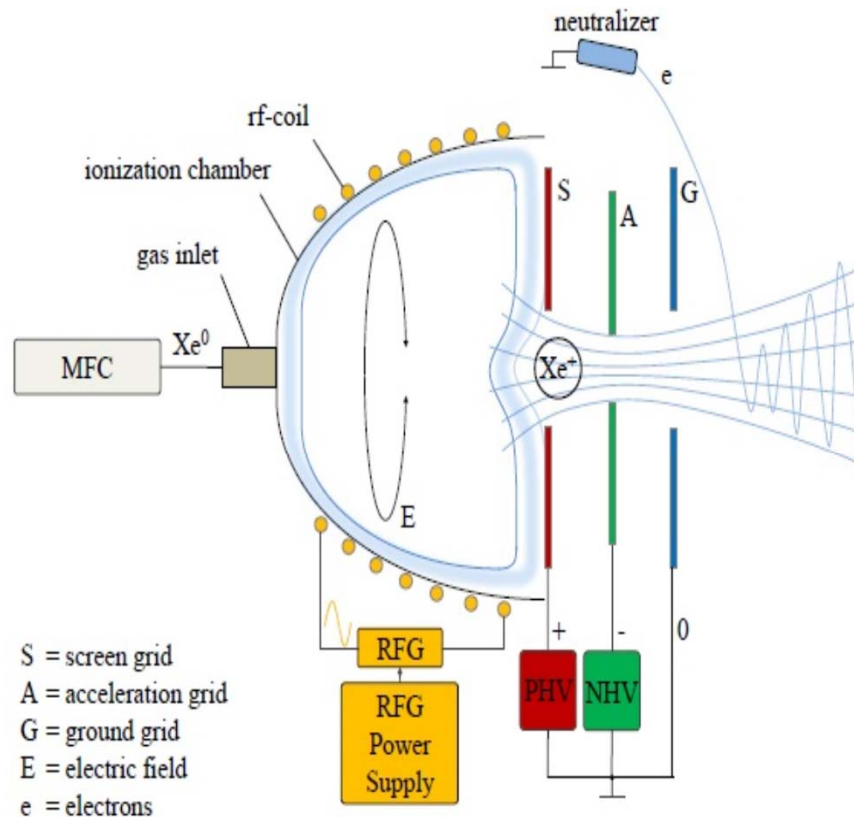


Mode1: sunlight enter the telescope,
temperature of M1
Difference between M1 and M2: 1°C



Mode2: no sunlight enter the telescope,
temperature of M1
Difference between M1 and M2: 2°C

Radio frequency ion thruster (μ RIT) for micropropulsion and satellite gravity In I.Mech

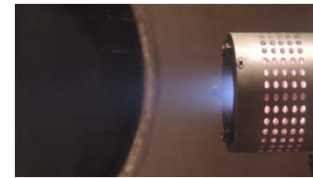
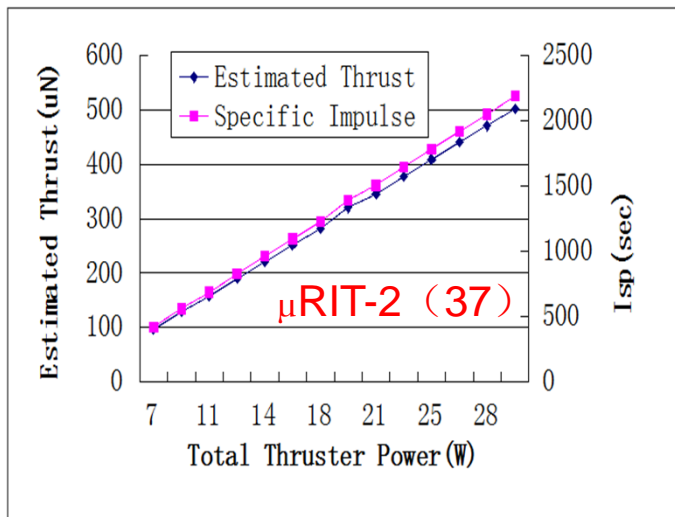
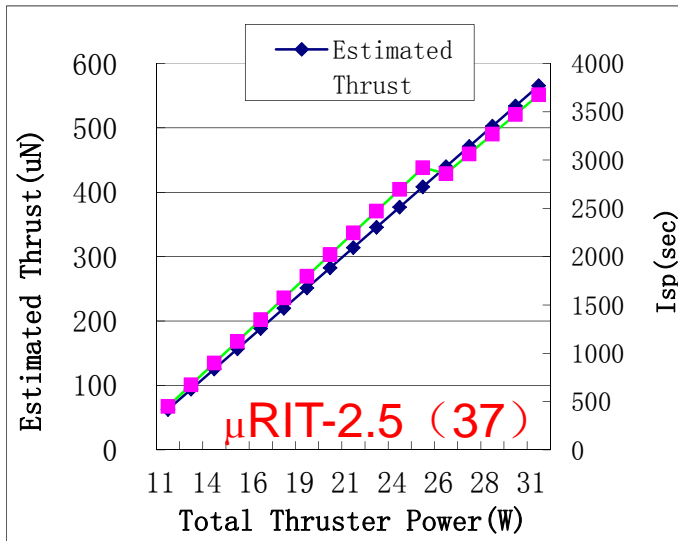


Function principle of a RIT

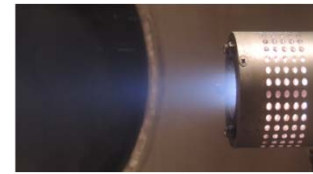
➤ RIT's advantages

- Cathodeless
- No magnets and high voltage biased devices
- No sputtered electrode
- High resolution and Low noise
- Fast response
- High thrust regulability

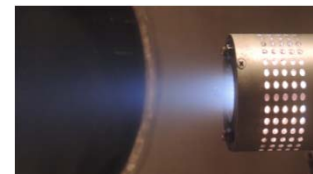
μRIT experimental results



~ 200 μN



~ 350 μN



~ 500 μN

μRIT-2(37)

Thruster Type	μRIT-2	μRIT-2.5
Thrust Range (μN)	100-500	63-565
Gas Flow Rate (μg/s)	21	16
Specific Impulse (s)	416-2190	450-3600
Specific Power (W/mN)	>60	>55
Propellant Utilization (%)	9-47	10-80
Energy Efficiency (%)	30-38.6	12-42

Launch Vehicle Technology

An escaped orbit launch service for the mission.



CZ-5 rocket will launch the main part of Chinese space station around 2020, Load: 25 tons.

CZ-5 is able to launch 3 satellites orbiting the Sun
Load: 5.7 tons

China Academy of Launch Vehicle Technology (CALT)

Mission for gravitational wave detection in Space in China

- Open for collaboration with international society, especially with ESA scientists;
- Develop and work with eLISA for collaboration and data comparison;
- Employ more technology in same models and types for high efficiency and low cost

International Scientific activities

- International conference on Gravitational Wave Detection in Space, Beijing, **April or May, 2017**
- **Taiji Union** for Gravitational Wave Detection in Space will be set up soon and all researchers and scientists all over the world are welcome.
- Joint Albert Einstein Institute for Radio Astronomy and Gravitational Physics between Max Planck Society and Chinese Academy of Sciences to be discussed in MPG-CAS Meeting to be held in Shanghai in **November, 2016**.
- **Chinese pathfinder** would be launched in 10 years to test key technology for GWD in Space or/and Grace II, International collaboration is welcome.
- **Joint research proposal** to be submitted to the DFG and National Science Foundation, China after the Sino-German meeting in 2015

Prospect

- Step I (2016-2020)

Ground studies on theoretical analyses and key technology

- Step II (2021-2025)

Space technology for Chinese pathfinder of key technology experiment

- Step III (2026 – 2035)

Satellite of SGW mission



Thank You!